

# Numerical Investigation of Shock Stand-off in Chlorine

---

**Robert Watt** and Rowan Gollan

10-07-2025

The University of Queensland

# Ballistic range experiments

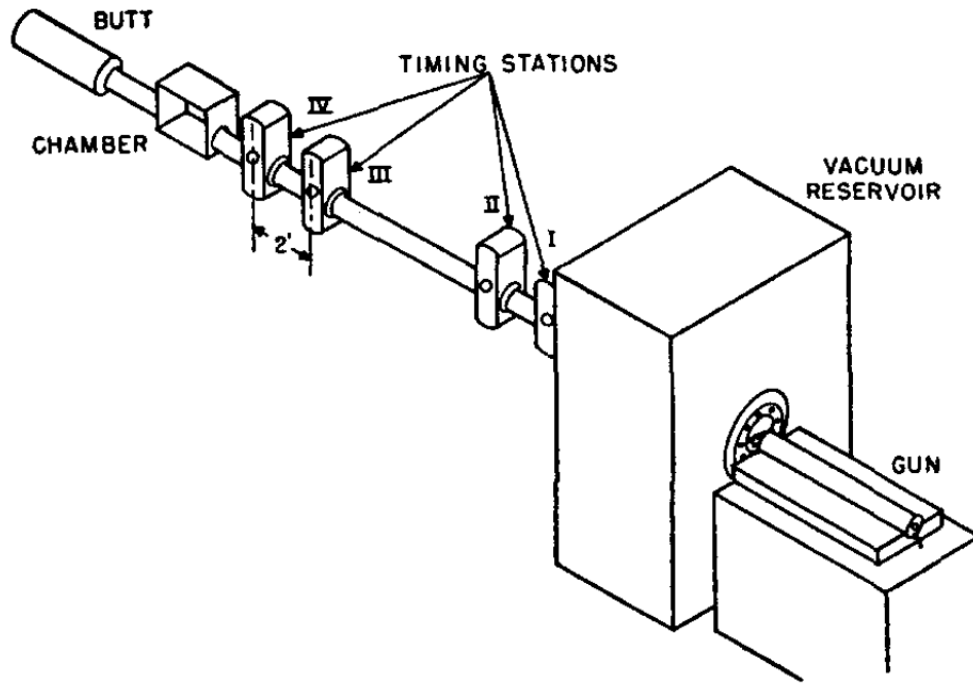
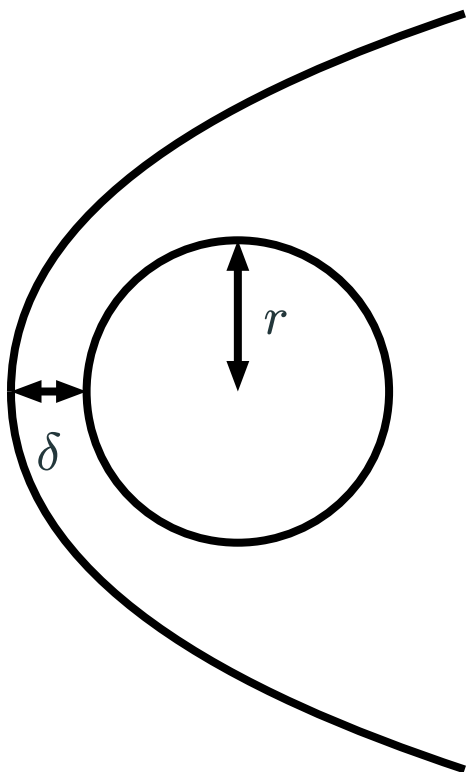


Figure 1: Schematic of ballistic range from Schwartz and Eckerman<sup>1</sup>

- Performed by Schwartz and Eckerman in 1956<sup>1</sup>
- Aim of experiment:
  - Measure vibrational relaxation of  $\text{Cl}_2$  by measuring shock stand-off distance
- Gun tunnel
  - 4 mm radius ball bearings, various materials
  - Schlieren/shadowgraphs to measure shock stand-off distance
- Hopefully good for CFD validation
  - Isolates vibrational relaxation (e.g. no dissociation, electronic excitation e.g.)

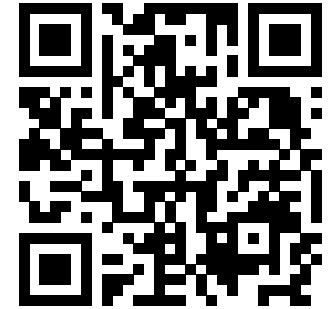
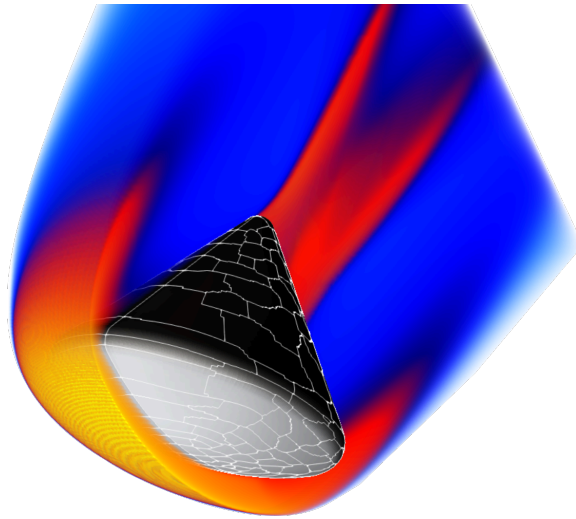
<sup>1</sup>Schwartz and Eckerman, 1956: *Shock Location in Front of a Sphere as a Measure of Real Gas Effects*



- Measurements of shock stand-off distance at various Mach numbers (2-4) and pressures ( $\sim 130$  Pa through  $\sim 25$  kPa)
- The level of vibrational nonequilibrium changes  $\gamma$ , which changes shock stand-off
- Matching shock stand-off in CFD requires getting vibrational relaxation correct

## Gas Dynamics Toolkit<sup>1</sup>

*GDTk is a collection of software for doing gas dynamics, from simple desktop calculations through to simulations on supercomputers*

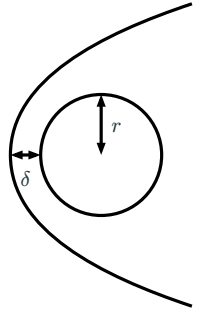
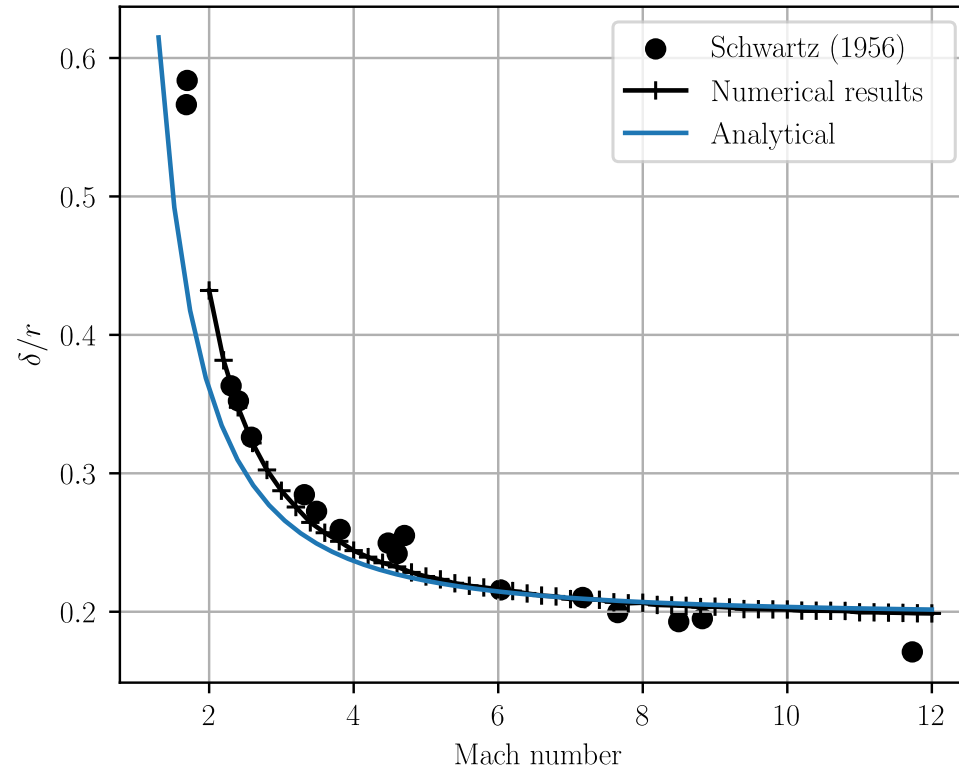


<https://gdtk.uqcloud.net>

Figure 2: Simulation of Apollo capsule with Eilmer

---

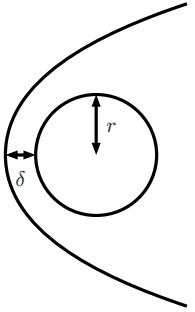
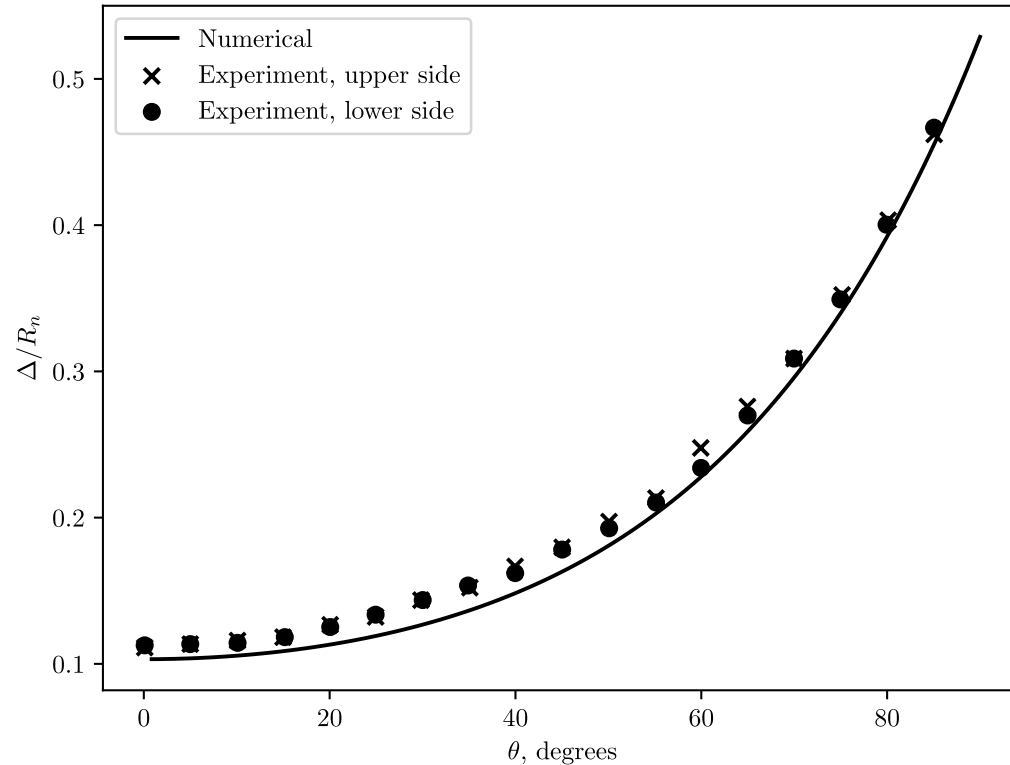
<sup>1</sup>Gibbons *et. al.*, 2023: *Eilmer: An open-source multi-physics hypersonic flow solver*



- Validation of the experimental setup
- Validation of the Eilmer<sup>1</sup>

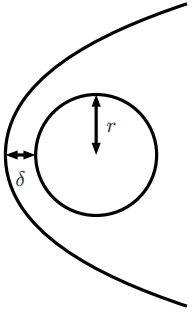
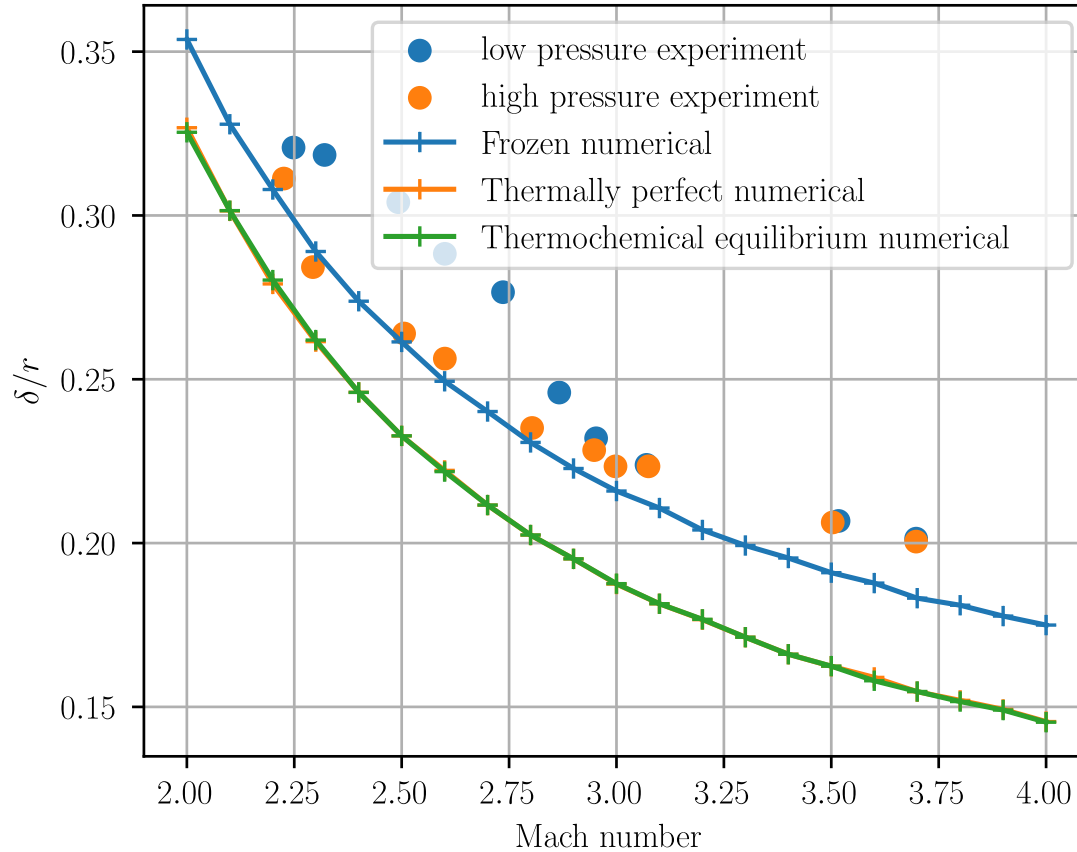
<sup>1</sup>Gollan and Jacobs, 2013: *About the formulation, verification and validation of the hypersonic flow solver Eilmer*

# Nonaka Sphere - Separate validation of Eilmer



- Shock shape validation with the Nonaka Sphere<sup>1</sup> in air
- Full thermochemical nonequilibrium is required to get the correct shock shape

<sup>1</sup>Nonaka *et al.*, 2012: *Measurement of shock stand-off distance for sphere in ballistic range*



- Results from over a decade ago
- Frozen and equilibrium assumptions do not bound experimental results

Figure 5: Chlorine shock stand-off distance

## Numerics

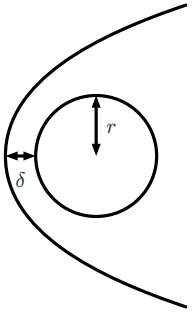
- Second-order finite-volume
- body-fitted grid
- Jacobian-Free Newton-Krylov for steady-state acceleration

## Gas Models

- Ideal  $\text{Cl}_2$
- Thermally perfect  $\text{Cl}_2$
- Equilibrium mixture of  $\text{Cl}_2$  and  $\text{Cl}$ 
  - Equilibrium computed with Nick Gibbons' eqc

## Simulations

- Half sphere, 2D axisymmetric
- Initial inviscid shock-fit solution
- Four successive viscous solutions on finer grids
  - Finest grid:  $210 \times 210$  with first cell  $1\mu\text{m}$
  - Grid is tailored to align with the shock of each simulation
- 20 Simulations from Mach 2-4, for each gas model





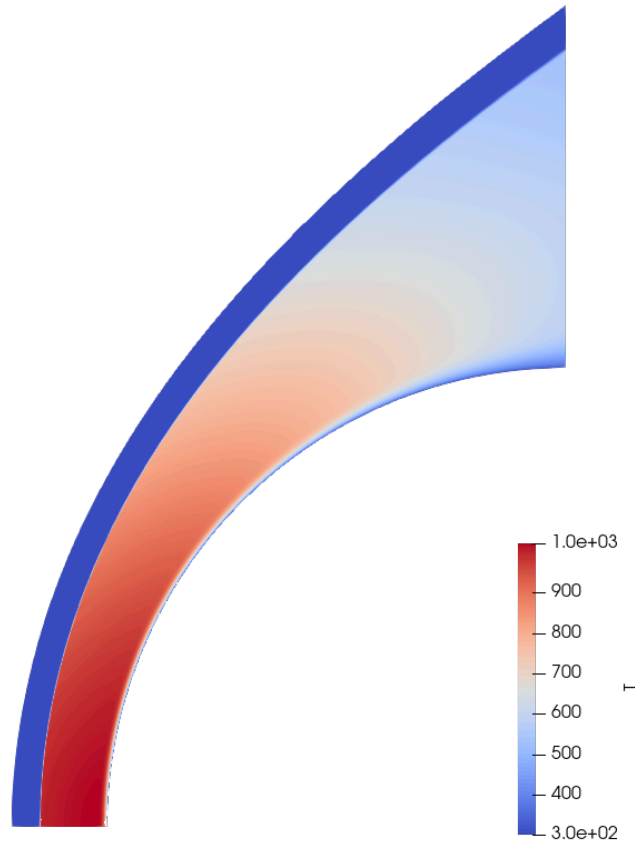


Figure 6: Representative flow field

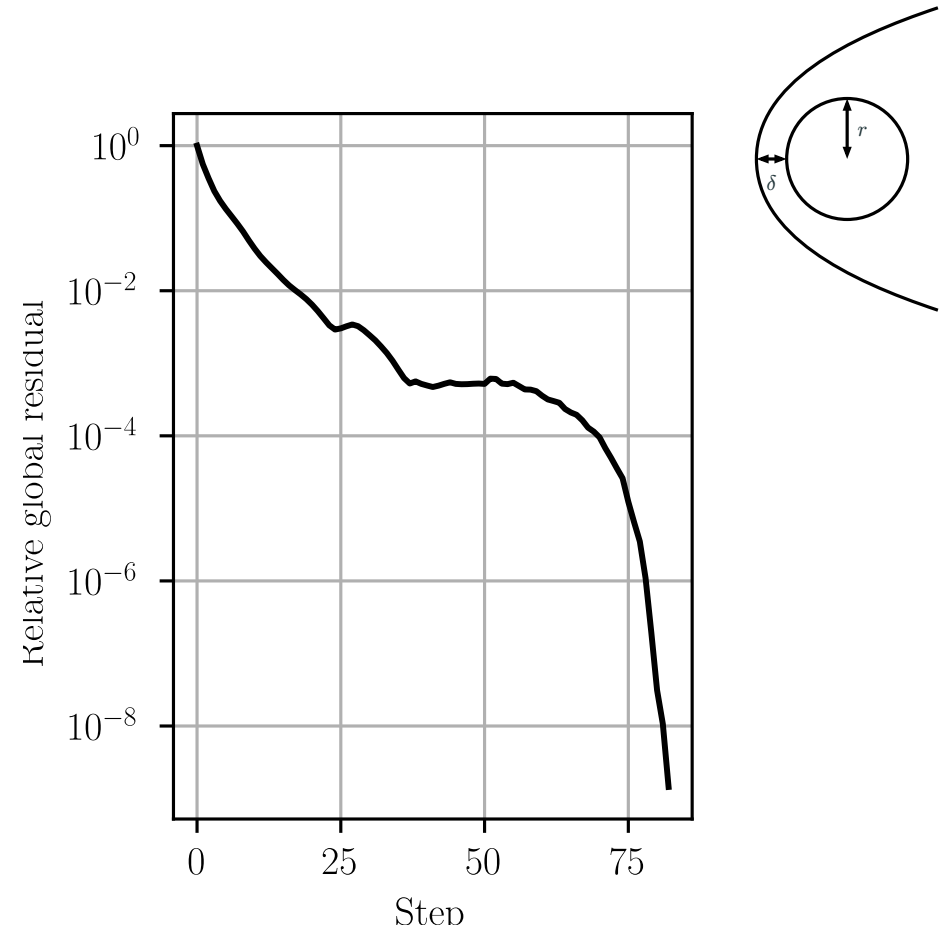
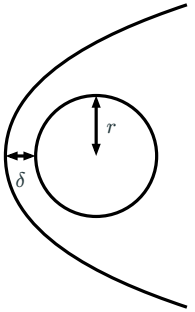
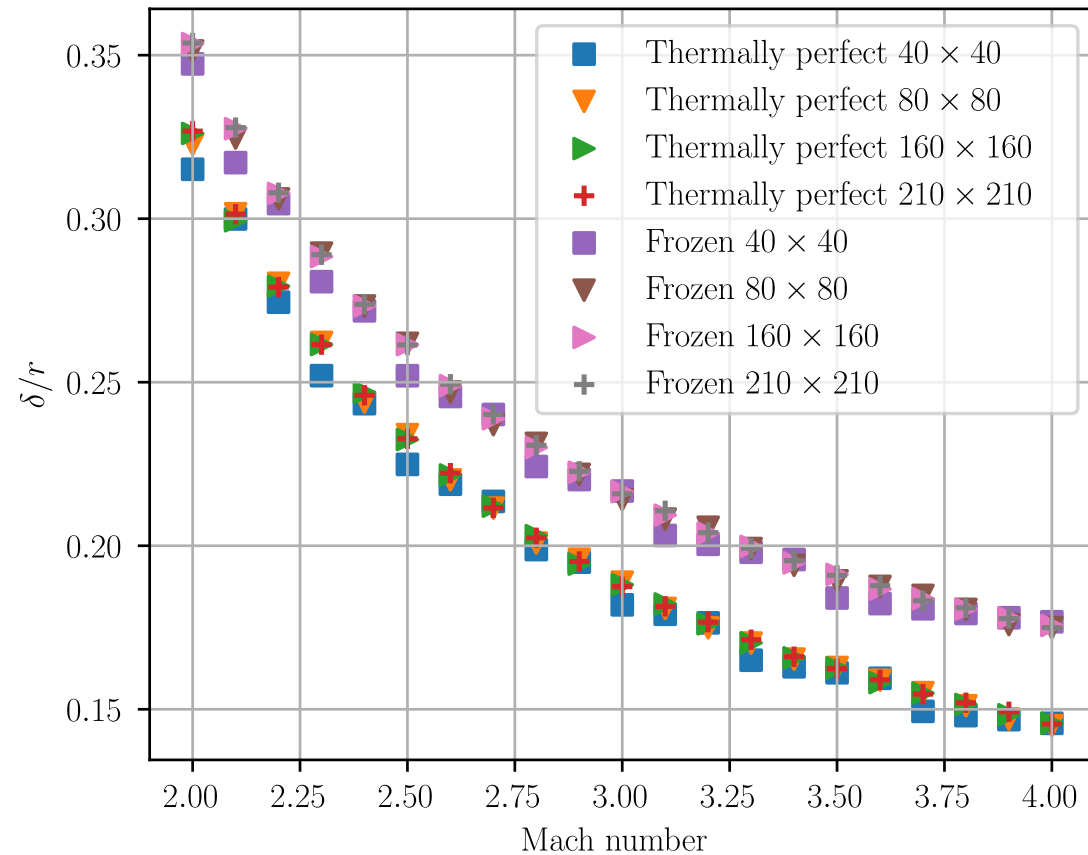
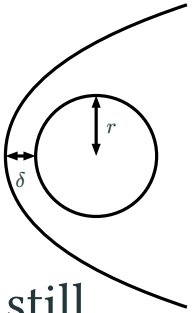
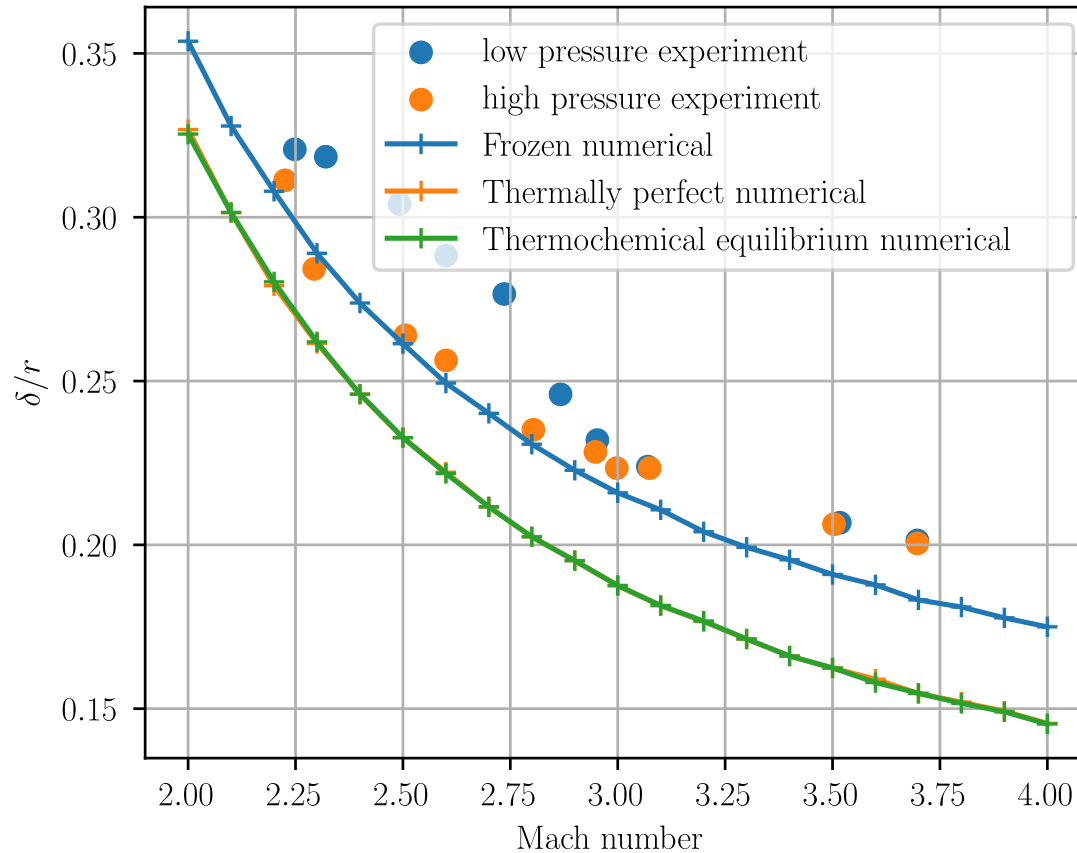


Figure 7: Representative convergence

# Grid Convergence



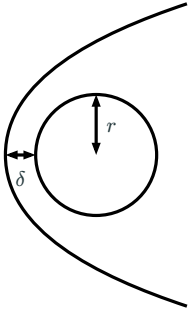
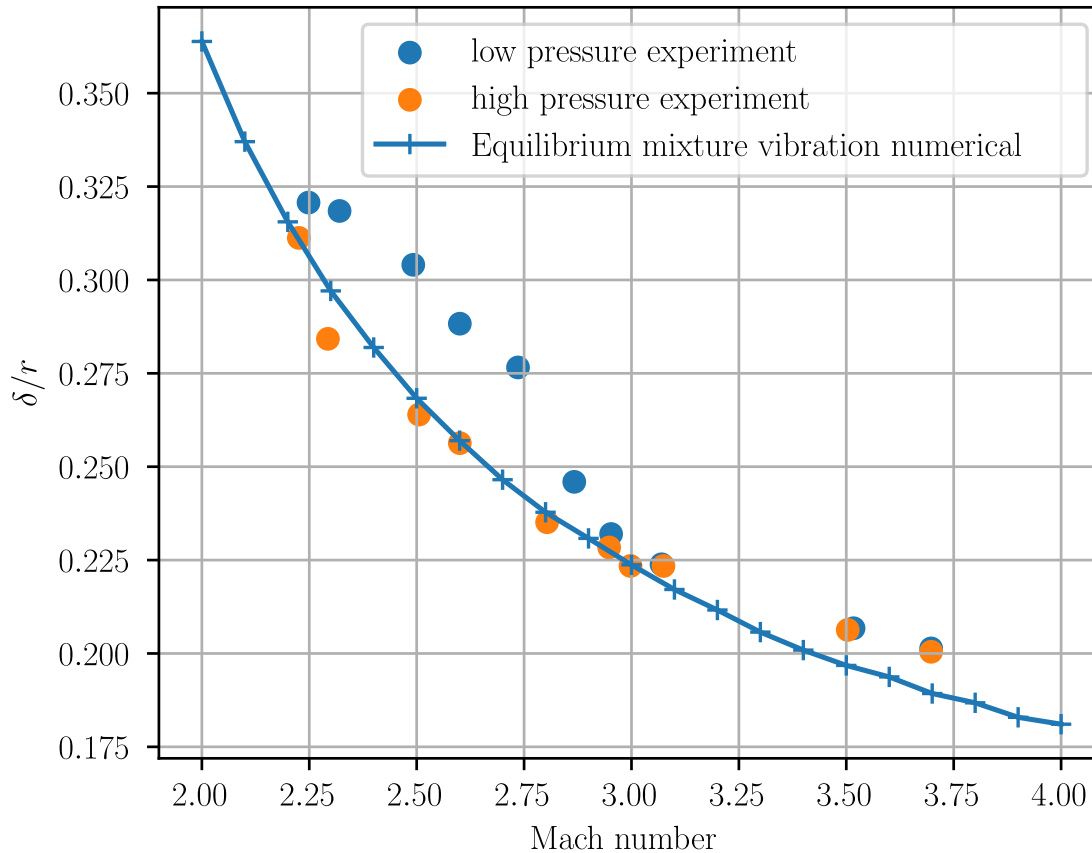
# Chlorine results



- Numerical shock stand-off is still too small
- Require *effective*  $\gamma$  to increase
  - Rules out  $\text{Cl}_2$  reacting and forming more complex molecules
  - Rules out contamination with oil or grease from the gun
- Or require speed to decrease

Figure 9: Comparison of simulated and observed shock stand-off

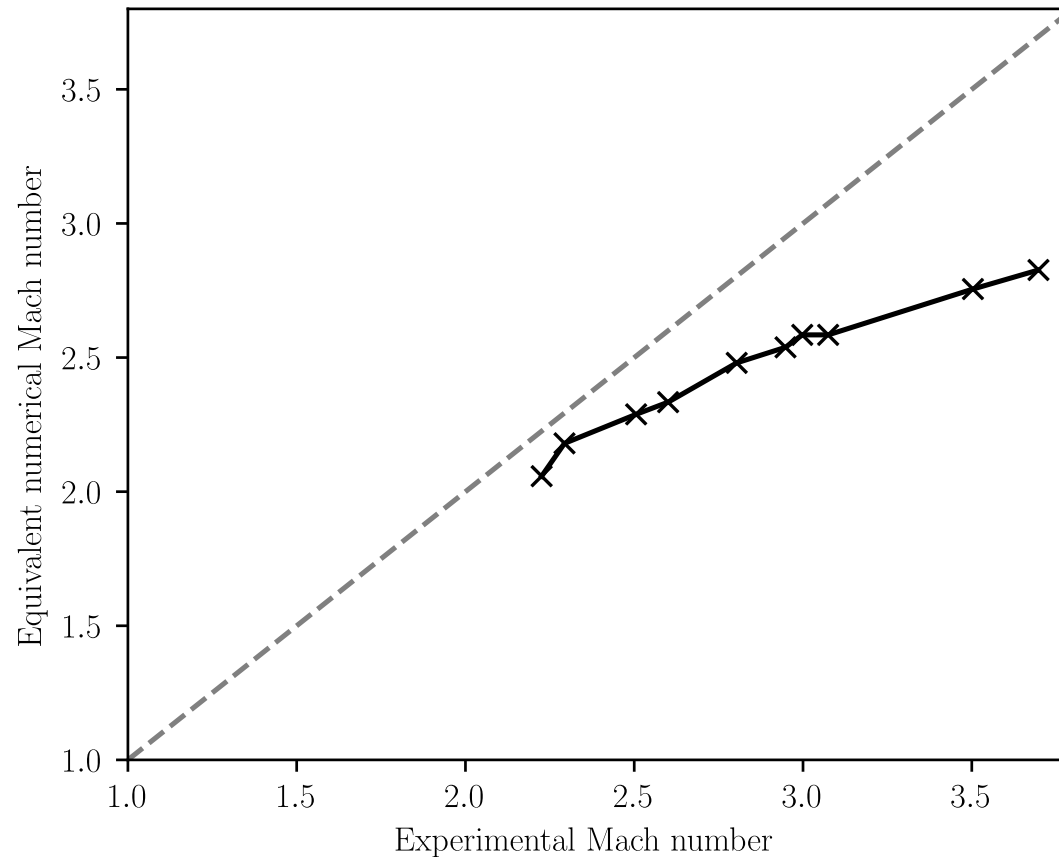
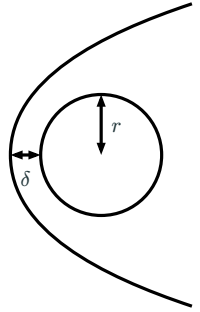
# Chlorine-Argon mixture



- Thermally perfect gas model
- 60%  $\text{Cl}_2$ , 40% Ar by mass
  - $\gamma_{\text{effective}} \approx 1.42$
- Unrealistic for there to be so much contamination

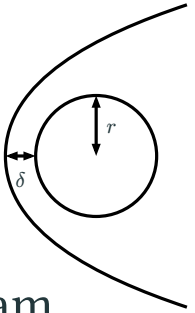
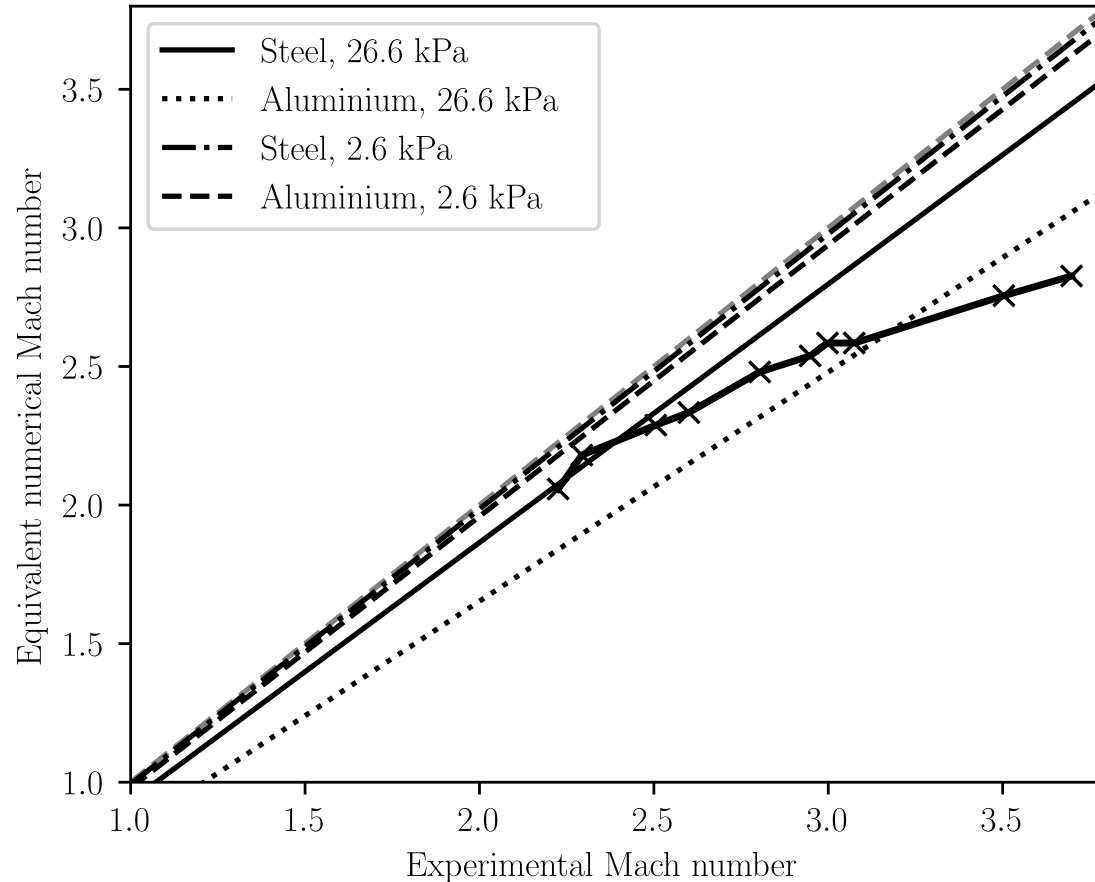
Figure 10: Mixture of Ar and  $\text{Cl}_2$

# Deceleration

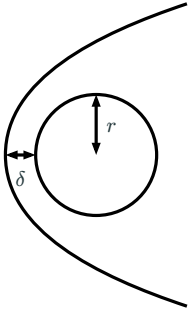


- Calculate the free-stream speed to get the same numerical shock stand-off as the experiment

# Deceleration



- Calculate the free-stream speed to get the same numerical shock stand-off as the experiment
- Analytic deceleration calculated based on steel and aluminium ball bearing, with a conservative estimate of  $C_D = 1.6$



- Unable to match the observed shock stand-off
- Explored some possible explanations for the differences
  - Viscous shock layer interactions ✗
  - Flow contamination ✗
  - Deceleration ?
- Not enough information in the paper to calculate deceleration (e.g. material of bearing, or pressure for each shot)
- Not enough information for validation of CFD codes